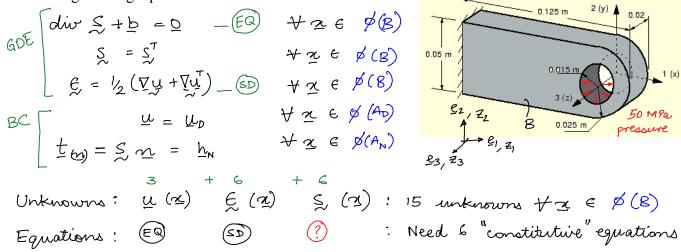
Chapter 4: Material Behavior

Recall governing equations:



Different materials behave differently when subjected to loads and deformation.

Thus stress-strain relationships depend upon material properties that have to be determined by conducting experiments in a lab (theory alone does not suffice).

General principles for material stress-strain (constitutive) relationships:

- Deterministic
- Local action
- Observer objectivity / Material frame indifference
- Laws of Thermodynamics

Types of material models:

- Elastic: Hyper-elastic, Hypo-elastic, Visco-elastic
- Inelastic: Plastic, Visco-plastic, Damage models ...

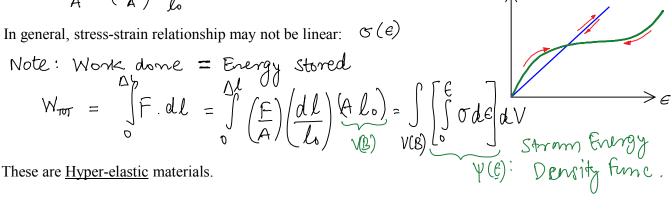
Elasticity

A fundamental tenet of elasticity is that when a loaded material (body) is unloaded, it returns to its original undeformed shape/state.

Another way of saying this is that stress-strain relationship at a point does <u>not</u> depend upon how / in what sequence of operations (history) the material point was loaded i.e. path-independence. It only depends on the current *state* of strain at the point.

1D Linear Hooke's law (model)
$$F = k \triangle l \quad ;$$

$$\frac{F}{A} = \left(\frac{k l_0}{A}\right) \frac{\Delta l}{l_0} \implies \sigma = C \epsilon$$



Strain energy (density) function

Stores the energy resulting from internal work done at each point of a body /structure.

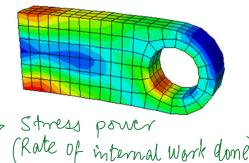
$$\Psi(\epsilon) = \int_{0}^{\epsilon} \sigma(\epsilon) d\epsilon \Rightarrow \sigma(\epsilon) = \frac{d\Psi}{d\epsilon}$$

If stress-strain relationship is linear (Hooke's model):

$$\psi(\varepsilon) = \frac{1}{2} C \varepsilon^2$$
 and $\sigma = C \varepsilon \Rightarrow C = \frac{\partial \sigma}{\partial \varepsilon} = \frac{\partial^2 \psi}{\partial \varepsilon^2}$

Generalization to 3D:

Stress and strain are tensors.



Thus $\dot{\psi} = \hat{S} : \dot{\epsilon}$ and noting $\dot{\psi} = \underbrace{\partial \psi}_{\partial \epsilon_{ij}} : \dot{\epsilon}_{ij}$ enam Rull

$$\Rightarrow \quad \mathcal{S} = \frac{\partial \Psi}{\partial \mathcal{E}} \quad \Rightarrow \quad \mathcal{S}\dot{y} = \frac{\partial \Psi}{\partial \mathcal{E}ij}$$

If stress-strain relationship is linear:

and

S = C & => Sij = Cijke Eke

4th order tensor (& elastic constants)

36: Symmetry of S; f.

21: Symmetry of L

$$C_{ij} \kappa \ell = \frac{\partial S_{ij}}{\partial \epsilon_{\kappa \ell}} = \frac{\partial^2 \psi}{\partial \epsilon_{ij} \partial \epsilon_{\kappa \ell}} ; \qquad \mathcal{C}_{\approx} = C_{ij} \kappa \ell \left(\underline{\ell} \otimes \underline{e}_{j} \otimes \underline{e}_{k} \otimes \underline{e}_{\ell} \right)$$

One can show: $\Psi(\xi) = \frac{1}{2} \epsilon_{ij} C_{ijke} \epsilon_{ke}$

Aside: 4th order Tensors:

If
$$T = [\underline{s} \otimes \underline{t} \otimes \underline{u} \otimes \underline{\varphi}]$$
; $S = (\underline{a} \otimes \underline{b})$
Then $T \subseteq [\underline{s} \otimes \underline{t}) \otimes (\underline{u} \otimes \underline{\varphi}]$ $(\underline{a} \otimes \underline{b}) = (\underline{u} \cdot \underline{a})(\underline{\varphi} \cdot \underline{b})$ $(\underline{s} \otimes \underline{t})$
Note $[(\underline{e}_i \otimes \underline{e}_j) \otimes (\underline{e}_k \otimes \underline{e}_k)][\underline{e}_m \otimes \underline{e}_m] = \delta_{km} \delta_{kn}$ $(\underline{e}_i \otimes \underline{e}_j)$
 $\Rightarrow Sij (\underline{e}_i \otimes \underline{e}_j) = Cijkl [(\underline{e}_i \otimes \underline{e}_j) \otimes (\underline{e}_k \otimes \underline{e}_k)] \in_{kn} (\underline{e}_m \otimes \underline{e}_m)$
Thus $Sij = Cijkl \in kl$

Voight notation

For convenience of visualization (and ease of computer implementation), the relationship between stress and strain is sometimes expressed in a different notation using matrices (Problem 101):

S_{11} S_{22} S_{33} S_{12} S_{23} S_{31} S_{21} S_{32}	$ \begin{array}{c} C_{1111} \\ C_{2211} \\ C_{2311} \\ C_{1211} \\ C_{2311} \\ C_{2311} \\ C_{2111} \\ C_{2111} \end{array} $	C _{11 22} C _{22 22} C ₃₃₂₂ C _{12 22} C _{23 22} C _{21 22} C _{21 22} C _{22 22}	C ₁₁₃₃ C ₂₂₃₃ C ₃₃₃₃ C ₁₂₃₃ C ₂₁₃₃ C ₂₁₃₃ C ₂₁₃₃ C ₂₁₃₃	C ₁₂₁₂ C ₂₃₁₂ C ₃₁₁₂ C ₂₁₁₂	C ₂₂₂₃ C ₂₂₃₁ C ₂₈₂₃ C ₂₈₃₁ C ₁₂₂₃ C ₁₂₃₁ C ₂₃₂₃ C ₂₃₃₁ C ₃₁₂₃ C ₃₁₃₁ C ₂₁₂₃ C ₂₁₃₁	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{c} \mathcal{C}_{11} \\ \mathcal{C}_{22} \\ \mathcal{C}_{33} \\ \hline \mathcal{C}_{12} \\ \mathcal{C}_{23} \\ \mathcal{C}_{31} \\ \mathcal{C}_{32} \end{array}$
\[\leq_{32} \] \[\leq_{13} \]	C3211 . C1311	C _{32 22} C ₁₃₂₂	C ₃₂₃₃ C ₁₃₃₃	G ₂₁₂ C ₁₃₁₂		3221 3252 3213	632 613

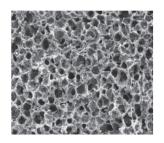
Taking advantage of symmetry as:

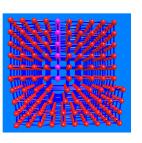
	Ū		U	U			_
Su		C_{1111}	C1122	C1133	C1112	C1123 C1131	ϵ_{ij}
522		C ₂₂₁₁	C2222	C2233	C ₂₂₁₂	C2223 C2231	€22
ح ₂₃	_	C ₂₃₁₁	C3322	C3333	C3312	C _{23 23} C ₃₃ C ₁₂₂₃	£33
512	_	C1211	C ₁₂₂₂	C1233	G ₂₁₂	C ₁₂₂₃ C ₍₂₃₎	<u>2</u> €12
S ₂₃		C2311				C2323 G331	2 € ₂₃
S ₃₁		C3111				C3123 C3131	$\stackrel{2}{\sim} \in_{3_1}$

Isotropy

Materials derive their properties from their internal micro-structure i.e. from the make up of their constituent atoms and molecules.

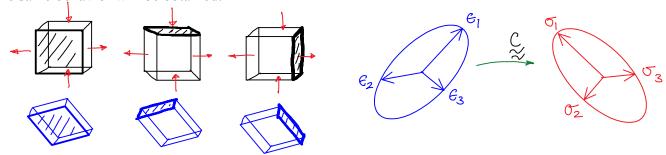
If this micro-structure is uniform and homogenous in all directions, then the material displays same behavior in all directions of loading and is called isotropic e.g. steel, plain concrete...





Note: Most materials are not isotropic e.g. wood, reinforced composites, natural materials...

If a piece of isotropic material is subjected to the same state of stress and strain, but in different orientations, the same behavior will be obtained.



One way to define isotropic hyper elastic materials is in terms of the invariants of strain.

An energy density function in terms of invariants is isotropic:

An energy density function in terms of invariants is isotropic:
$$\frac{1}{2} \left(\frac{1}{2} \left(\frac{1}{2} \right) \right) + \frac{1}{2} \left(\frac{1}{2} \right)$$

Thus
$$S_{r} = \frac{\partial \psi}{\partial f_{1}} \cdot \frac{1}{2} + \frac{\partial \psi}{\partial f_{2}} \cdot \frac{1}{2} + \frac{\partial \psi}{\partial f_{3}} \cdot \frac{1}{2} + \frac{\partial \psi}{\partial f_{3}}$$

Linear Isotropic Hyper elastic material:

Strain energy density function must be quadratic:

$$\Psi(\xi) = \alpha_1 \left[\operatorname{tr}(\xi) \right]^2 + \alpha_2 \operatorname{tr}(\xi^2)$$

A particular form of this is:
$$\psi(\xi) = \frac{\lambda}{2} [tr(\xi)]^2 + u tr(\xi^2)$$

$$\Rightarrow \lesssim = \frac{\partial \psi}{\partial f_1} \cdot \overline{\lambda} + \frac{\partial \psi}{\partial f_2} \cdot 2 \frac{6}{2} \Rightarrow \lesssim = \lambda \operatorname{tr}(e) \overline{\lambda} + 2 \mu \frac{6}{2}$$

$$Lamé parameters: \lambda, \mu$$
(2 constants)

This is the 3D version of the Hooke' (law) model.

Note: Cijke =
$$\frac{\partial Sij}{\partial \mathcal{E}_{KR}} = \frac{\partial \left(\lambda \in_{mm} \delta_{ij} + 2u \in ij\right)}{\partial \mathcal{E}_{KR}}$$

= $\lambda S_{MK} S_{MR} S_{ij} + 2u S_{iK} S_{jL}$
Alternatively

Cijke = $\lambda S_{ij} S_{KR} + u \left[S_{iK} S_{jR} + S_{iL} S_{jK}\right]$
Using this we can express: $S_{ij} = C_{ijk} \in \mathcal{E}_{KR}$

Another useful form: (given stress, find strain):

$$\mathcal{E}_{S} = -\frac{\lambda}{2\pi(3\lambda + 2\pi)} \operatorname{tr}(S) I + I S$$

Finding Material Constants Experimentally: 1: Uniaxial tension

Recall Uniaxial Tension:
$$S = O_1(t_1 \otimes t_1) \times O_1 \circ O_2$$
; $C \sim C_1 \circ C_2 \circ C_3 \circ C_3$



$$-\underline{t}_{(\underline{e_1})} - (\underline{s}\underline{e_1}) + \underline{t}_{(\underline{e_1})} = \underline{s}\underline{e_1}$$

Solving for strains using the Hooke's model:

$$\xi = \frac{-\lambda}{2\pi(3\lambda + 2\pi)} \sigma_1 \vec{1} + \frac{1}{2\pi} \vec{S} \Rightarrow \xi_1 = \xi_1 = \frac{(\lambda + \pi)}{\pi(3\lambda + 2\pi)} \sigma_1$$

$$\Rightarrow$$
 Young's Modulus: $C = \frac{\sigma_1}{e_1} = \frac{\mu(3\lambda + 2\pi)}{(\lambda + \mu)}$ $\begin{cases} \text{Note:} \\ e_1 = \frac{\partial u_1}{\partial z_1} \approx \frac{\Delta l}{l_0} \end{cases}$

$$\begin{cases} \text{Note:} \\ \mathcal{E}_1 = \frac{\partial u_1}{\partial z_1} \approx \frac{\Delta l}{l_0} \end{cases}$$

Similarly
$$\epsilon_{22} = \epsilon_{33} = \frac{-\lambda}{2\mu(3\lambda + 2\mu)} = \frac{-\lambda}{2(\lambda + \mu)} \epsilon_{11}$$

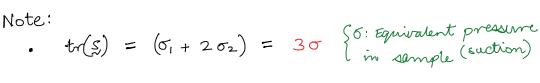
$$\Rightarrow$$
 Poisson's Ratio: $\gamma = \frac{-\epsilon_{22}}{\epsilon_{11}} = \frac{-\epsilon_{33}}{\epsilon_{11}} = \frac{\lambda}{2(\lambda + \mu)}$

Finding Material Constants Experimentally: 2: Tri-axial test

For a triaxid test:

$$S = \sigma_1 \left(\underbrace{e_1 \otimes e_1} \right) + \sigma_2 \left(\underbrace{e_2 \otimes e_2} \right) + \underbrace{e_3 \otimes e_3} \right]$$

$$E = E_1 \left(\underbrace{e_1 \otimes e_1} \right) + E_2 \left[\left(\underbrace{e_2 \otimes e_2} \right) + \left(\underbrace{e_3 \otimes e_3} \right) \right]$$



•
$$tr(\underline{\epsilon}) = (\epsilon_1 + 2\epsilon_2) = \epsilon$$
 {e: Dilatation

• Change in volume:
$$\frac{U}{V} = \det(\overline{F}) = \sqrt{\det(\overline{G})} = \sqrt{\det(\overline{I} + 2\overline{E})}$$

One can show that $\underline{U} = \det(\overline{F}) \approx 1 + \operatorname{tr}(\overline{E}) \approx 1 + \operatorname{tr}(\overline{E})$

$$\Rightarrow$$
 Dilatation: $e = \frac{\Delta V}{V} = \frac{\omega - V}{V} = det(E) - 1 \approx tr(E)$

From the Hooke's model:
$$tr(S) = \lambda tr(E) tr(I) + 2u tr(E)$$

$$\Rightarrow tr(S) = (3\lambda + 2u) tr(E)$$

$$\Rightarrow \sigma = 3k$$

$$\Rightarrow \sigma = ke$$
Bulk modulus
$$\begin{cases} k = \sigma = (\lambda + \frac{2}{3}u) \\ = 0 \end{cases}$$

Define Deviatoric Stress:
$$S' = S - \frac{1}{3} tr(S)I = S - \sigma I$$
 { Note: $t_{\sigma}(S') = 0$ Again from Hooke's model:

$$S' = \lambda \operatorname{tr}(\underline{\epsilon}) \overline{1} + 2u(\underline{\epsilon}) - \sigma \overline{1} = 2u(\underline{\epsilon} - \frac{1}{3} \operatorname{tr}(\underline{\epsilon}) \overline{1})$$
 $\underline{\epsilon}' = 2u(\underline{\epsilon}) \overline{1} + 2u(\underline{\epsilon}) - \sigma \overline{1} = 2u(\underline{\epsilon} - \frac{1}{3} \operatorname{tr}(\underline{\epsilon}) \overline{1})$

Thus S' = 2m &'

In terms of
$$K, \mathcal{U}: \begin{cases} S = S' - \frac{1}{3} \operatorname{tr}(S) I = K \operatorname{tr}(E) I + 2 \mathcal{U} E' \end{cases}$$

Thus by measuring
$$\sigma_1$$
, σ_2 , e , and ϵ_1 we can find
$$K = \frac{\sigma}{e} = \frac{(\sigma_1 + 2\sigma_2)}{3e} \quad ; \quad \epsilon_2 = \frac{e - \epsilon_1}{2}$$

Noting that
$$S' = \frac{1}{3}(\sigma_1 - \sigma_2) \left[3(\underline{e}_1 \otimes \underline{e}_1) - \underline{I}\right] \Rightarrow 2u = \frac{\sigma_1 - \sigma_2}{\varepsilon_1 - \varepsilon_2}$$
 and $\varepsilon' = \frac{1}{3}(\varepsilon_1 - \varepsilon_2) \left[3(\underline{e}_1 \otimes \underline{e}_1) - \underline{I}\right]$

<u>Useful relationships between material moduli</u> (Problem 98)

All the material moduli are related and can be derived in terms of one another.

For a linear hyper-elastic isotropic material one needs any two of these material constants to describe the relationship between in stress and strain.

$$\lambda \ = \ \frac{2\mu\nu}{1-2\nu} \ = \ \frac{\mu(C-2\mu)}{3\mu-C} \ = \ \frac{C\nu}{(1+\nu)(1-2\nu)} \ = \ \frac{3K\nu}{1+\nu}$$

$$K = \lambda + \frac{2}{3}\mu = \frac{\mu C}{3(3\mu - C)} = \frac{\lambda(1+\nu)}{3\nu} = \frac{C}{3(1-2\nu)}$$

$$C = 2\mu(1+\nu) = \frac{\mu(3\lambda+2\mu)}{\lambda+\mu} = \frac{\lambda(1+\nu)(1-2\nu)}{\nu} = \frac{9K\mu}{3K+\mu}$$

$$\mu = \frac{C}{2(1+\nu)} = \frac{3}{2}(K-\lambda) = \frac{3K(1-2\nu)}{2(1+\nu)} = \frac{\lambda(1-2\nu)}{2\nu}$$

$$v = \frac{\lambda}{2(\lambda + \mu)} = \frac{C}{2\mu} - 1 = \frac{3K - 2\mu}{2(3K - \mu)} = \frac{3K - C}{6K}$$

Note:

· Under uniavial tension $\sigma_1 = C \epsilon_1$

• Under hydrostatic pressure state of stress S = -PI $\Rightarrow \frac{1}{2}tr(S) = -P = \sigma$

Dilatation:
$$e < 0 \Rightarrow \frac{\sigma}{\kappa} < 0 \Rightarrow \frac{-P}{\kappa} < 0 \Rightarrow \kappa > 0$$

For Bulk modulus K70 => 1-2270 => 25/2

Further since the Lamé parameter λ can be negative, $K = \lambda + \frac{2}{3}M \Rightarrow M > 0 \Rightarrow 1 + \nu > 0 \Rightarrow \nu > -1$

Thus limits on Poisson's ratio: -1< 2 < 1/2

· Lamé parameter u is also referred to as "G" snears

Hyper-elasticity for finite (large) strains:

Let's revisit the Strain Energy density function:

In spatial (deformed) configuration:

External Work done (over incremental displacement du:

$$W_{TOT} = \int d\underline{u}(\underline{x}) \cdot \underline{t}_{(\underline{n})} da + \int d\underline{u}(\underline{x}) \cdot \underline{b} dv$$

$$\emptyset(\underline{B})$$

Note: $du \cdot t_{(1)} = du \cdot s_n = (s_du) \cdot n$

Using Divergence Theorem: $\int (S, du) \cdot n da = \int div (S, du) dv$

Also note: $div(\underline{S}, d\underline{u}) = div(\underline{S}) \cdot d\underline{u} + \underline{S} : \underline{\nabla}_{\underline{z}}(d\underline{u})$

Thus $W_{TOT} = \int d\underline{u} \cdot (d\underline{u} \cdot \underline{S} + \underline{b}) d\underline{u} + \int \underline{S} : \underline{\nabla}_{\underline{x}} \frac{d\underline{u}}{dt} dt d\underline{u}$

Over the entire deformation process:

 $W_{\text{ror}} = \iint_{\emptyset(B)} \underbrace{\mathbb{S}} : \underbrace{\mathbb{L}} dt dt dv = \iint_{B} \underbrace{\mathbb{S}} : \underbrace{\mathbb{L}} dt dt dv = \underbrace{\mathbb{E}}_{E}^{T} dv dv = \underbrace{\mathbb{E}}_{E}^{T} dv dv dt$

It can be shown that $W_{TOT} = \int_{B}^{T} \left[\int_{B} (J S F^{-T}) : \dot{F} dV \right] dt$ and finally:

 $W_{TOT} = \int \left[\int_{\mathcal{B}}^{t} \mathbf{S} : \mathbf{L} \, d\mathbf{v} \right] dt = \int_{0}^{t} \left[\int_{\mathcal{B}} \mathbf{P} : \dot{\mathbf{F}} \, d\mathbf{v} \right] dt = \int_{0}^{t} \left[\int_{\mathcal{B}} \mathbf{S} : \dot{\mathbf{F}} \, d\mathbf{v} \right] dt$

Thus

Cauchy Stress S is work (rate) conjugate to velocity gradient to 1st Piola Kirchhoff stress P " to rate of deformation gradient € 2nd Piola Kirchhoff stress S " to rate of Green-Lagrange strain €

When $=\int_{0}^{t} \left[\int_{\mathbb{R}} \dot{\psi}(\nabla_{\underline{x}}(d\underline{u})) d\underline{u}\right] dt = \int_{\mathbb{R}} \left[\int_{0}^{t} \dot{\psi}(\underline{F}) dV\right] dt = \int_{\mathbb{R}} \left[\int_{0}^{t} \dot{\overline{\psi}}(\underline{F}) dV\right] dt$ $=\int_{0}^{t} \left[\int_{\mathbb{R}} \dot{\psi}(\underline{F}) dV\right] dt = \int_{\mathbb{R}} \left[\int_{0}^{t} \dot{\overline{\psi}}(\underline{F}) dV\right] dt = \int_{\mathbb{R}} \left[\int_{0}^{t} \dot{\overline{\psi}}(\underline{F}) dV\right] dt$

Examples of Isotropic Hyper-elastic Models:

Note that
$$\overline{\Psi}(E) = \overline{\Psi}(C)$$
 such that $\dot{\lesssim} = \frac{\partial \overline{\Psi}}{\partial E} = 2 \frac{\partial \overline{\Psi}}{\partial C}$

• Mooney-Rivlin model:
$$\bar{\Psi}(\mathcal{C}) = a(I_c-3) + b(I_c-3)$$

• St. Venant Kirchhoff model:
$$\bar{\varphi}(\bar{\xi}) = \frac{\lambda}{2} \operatorname{tr}(\bar{\xi})^2 + u \operatorname{tr}(\bar{\xi}^2)$$

(or improved) : $\bar{\varphi}(\bar{\xi}) = \frac{\lambda}{2} (\ln \tau)^2 + u \operatorname{tr}(\bar{\xi}^2)$

• Neo-Hookean model (Compressible):
$$\bar{\Psi}(\mathcal{L}) = \frac{\mu}{2} \left(I_c - 3 \right) - \mu \ln J + \frac{\mu}{2} \left(\ln J \right)^2$$

Note: Perfectly incompressible (and nearly incompressible) materials require special treatment to ensure det (F) = J = 1