

HW7 GPS pseudorange Adjustment (26 Nov 2011)

④ Single epoch 13:00:00

$n = 8$ Indirect Observations

$n_0 = 4$ parameters: X_R, Y_R, Z_R, dt_R

$r = 4$ $F: PR_{\text{corr}} - [(X_R - X_s)^2 + (Y_R - Y_s)^2 + (Z_R - Z_s)^2]^{1/2} - c \cdot dt_R = 0$
 $\sigma_0 = 25 \text{ m } (.025 \text{ km!})$

Results:

no satellites rejected due to 10° elevation angle mask

Receiver coordinates:

residuals: .0159 (km)

$\hat{X}_R = 260.633235 \text{ (km)}$

- .10078

.10323

$\hat{Y}_R = -4853.044990$

.0197

.0016

$\hat{Z}_R = 4116.971011$

- .10318

- .0243

- .0056

global test: $\chi^2_* = 5.43$, $CV_1 = 0.48$, $CV_2 = 11.14$
 (2-sided) \Rightarrow Pass (accept H_0)

⑤ Three epochs 13:00:00, 13:30:00, 14:00:00

$n = 24$ (1 satellite rejected due to elev. angle $< 10^\circ$
 sat 28 in epoch 3)

$n_0 = 6$ Indirect Observations

$r = 18$ parameters: $X_R, Y_R, Z_R, dt_{R_1}, dt_{R_2}, dt_{R_3}$

results:

residuals:

$\hat{X}_R = 260.639064 \text{ (km)}$

.0207

.0151

.0175

.0028

.0076

.0231

$\hat{Y}_R = -4853.022217$

.0331

.0355

- .0572

.0112

.0107

.0457

$\hat{Z}_R = 4116.937166$

.0006

.0039

.0161

- .0281

- .0347

- .0305

- .0357

- .0357

- .0231

- .0046

- .0024

.0083

$RMS_v = .0257 \text{ km}$

global test: $\chi^2^* = 25.49$, $CV_1 = 8.23$, $CV_2 = 31.52$
 (2-sided) \Rightarrow pass

⑥ epoch 13:00:00, receiver constrained to be 100m from
 $(260.679818, -4852.970302, 4116.969922) \leftarrow (X_c, Y_c, Z_c)$

$n = 8$ $u = 4$: X_R, Y_R, Z_R, dt_R
 $n_o = 3$ $s = 1$: one constraint equation
 $r = 5$ $c = 8$: indirect observations

$$c + s \stackrel{?}{=} r + u \quad F_c: 0.100 - [(X_R - X_c)^2 + (Y_R - Y_c)^2 + (Z_R - Z_c)^2]^{1/2} = 0$$

$$8 + 1 = 5 + 4$$

$$9 = 9 \checkmark$$

results:

$$\hat{X}_R = 260.629524$$

$$\hat{Y}_R = -4853.056171$$

$$\hat{Z}_R = 4116.979770$$

residuals:

$$.0170$$

$$-.0123$$

$$.0296$$

$$.0225$$

$$.0037$$

$$-.0324$$

$$-.0226$$

$$-.0055$$

global test: $\chi^2^* = 5.50$, $CV_1 = 0.83$, $CV_2 = 12.83$
 (2-sided) \Rightarrow pass

$$F_c = -1.3 \times 10^{-6}$$

in listings: ④ gps-nav51a*

⑤ gps-nav53a*

⑥ gps-nav51b*

check for elevation angle :

$$\begin{bmatrix} e \\ h \\ \mu \end{bmatrix}_S = M_x(90^\circ - \phi) \cdot M_z(\lambda + 90^\circ) \left[\begin{bmatrix} X_S \\ Y_S \\ Z_S \end{bmatrix} - \begin{bmatrix} X_R \\ Y_R \\ Z_R \end{bmatrix} \right]$$

$\phi, \lambda \neq X_R Y_R Z_R$ all refer to receiver location

$$\begin{pmatrix} X \\ Y \\ Z \end{pmatrix} = \begin{bmatrix} (N+h) \cos \phi \cos \lambda \\ (N+h) \cos \phi \sin \lambda \\ ((1-e^2)N+h) \sin \phi \end{bmatrix}$$

$\phi, \lambda, h \rightarrow XYZ$
closed form

$$N = a / (1 - e^2 \sin^2 \phi)^{1/2}$$

$$a = 6378137.0 \text{ m}$$

$$b^2 = a^2 (1 - e^2)$$

$$1/f = 298.257223563$$

$$e^2 = \frac{a^2 - b^2}{a^2}$$

$$e^2 = 2f - f^2$$

$$f = \frac{a - b}{a}$$

$XYZ \rightarrow \phi \lambda h$ iterative solution

approximation for ϕ

loop
until
no
change

→ compute N

$$h = \frac{(X^2 + Y^2)^{1/2}}{\cos \phi} - N$$

$$\phi = \tan^{-1} \left[\frac{Z}{(X^2 + Y^2)^{1/2}} \left(1 - e^2 \left(\frac{N}{N+h} \right) \right)^{-1} \right]$$

$$\lambda = \tan^{-1} \left(\frac{Y}{X} \right)$$

use 2 argument arctangent function
 $\text{atan2}(Y, X)$

gps_nav51a

gps_nav51a

numsat =

8

sats =

16

13

23

10

6

7

19

3

m =

8

numsat =

8

after editing out defective sats

numsat =

8

c1 xs ys zs dt

ans =

Columns 1 through 3

21287141.34	9671.17665	-11273.286067
20287554.48	-5734.807399	-20258.741892
20605103.28	-1126.552213	-25140.371937
24293518.04	-22017.451052	-5852.11655
21364948.06	14140.406396	-14804.666272
22499418.9	-15186.548693	-8878.439153
21662186.34	8347.789114	-24844.425342
20253866.26	11027.953133	-19286.137311

Columns 4 through 5

22027.954669	-200.884996
16005.394146	248.828766
7915.839784	271.527222
13742.241308	-14.912353
16924.338317	-58.414824
19985.16409	29.497196
2801.519349	-189.641841
13886.297212	775.502939

sat. id

ans =

1 16

east, north, up of satellite for elev. angle check

ans =

9052.76533048471	9140.70388902187	16884.2719688828
------------------	------------------	------------------

elevd =

52.694616383131

sat. id

ans =

2 13

east, north, up of satellite for elev. angle check

ans =

-6812.87323234033	-727.888551905959	19174.7724344859
-------------------	-------------------	------------------

elevd =

70.336864806318

sat. id

ans =

3 23

east, north, up of satellite for elev. angle check

ans =

-2473.01125170978	-10206.5150230189	17822.5684034722
-------------------	-------------------	------------------

elevd =

gps_nav51a

```

59.4915299610946
# sat. id
ans =
  4 10
east, north, up of satellite for elev. angle check
ans =
  -22299.577380379          7451.52569989327          6095.92019233292
elevd =
  14.5351729699529
# sat. id
ans =
  5 6
east, north, up of satellite for elev. angle check
ans =
  13326.2051335967          2813.76579120993          16438.0353737534
elevd =
  50.3559565824672
# sat. id
ans =
  6 7
east, north, up of satellite for elev. angle check
ans =
  -15640.7804486962          10003.0547395423          12724.9520507625
elevd =
  34.4264003216796
# sat. id
ans =
  7 19
east, north, up of satellite for elev. angle check
ans =
  7003.56861856772          -14235.8109717316          14665.6379957201
elevd =
  42.749785202393
# sat. id
ans =
  8 3
east, north, up of satellite for elev. angle check
ans =
  9977.9238214494          -2293.3252966074          17744.6625657583
elevd =
  60.0164812847079
condN =
  188.244407355102
disp_del =
  1
  0.0292350003327373
  -0.0269903061604451
  0.0450107887463371
  0.121608840543441
condN =
  188.242825279625
disp_del =
  2
  -4.77171431272408e-008
  -4.87734835879411e-008
  1.54302779441388e-008
  -2.90479135596788e-008
we have converged
receiver location (km)
ans =
  260.633234952616          -4853.04499035493          4116.97101080418
receiver clock bias (usec)

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```

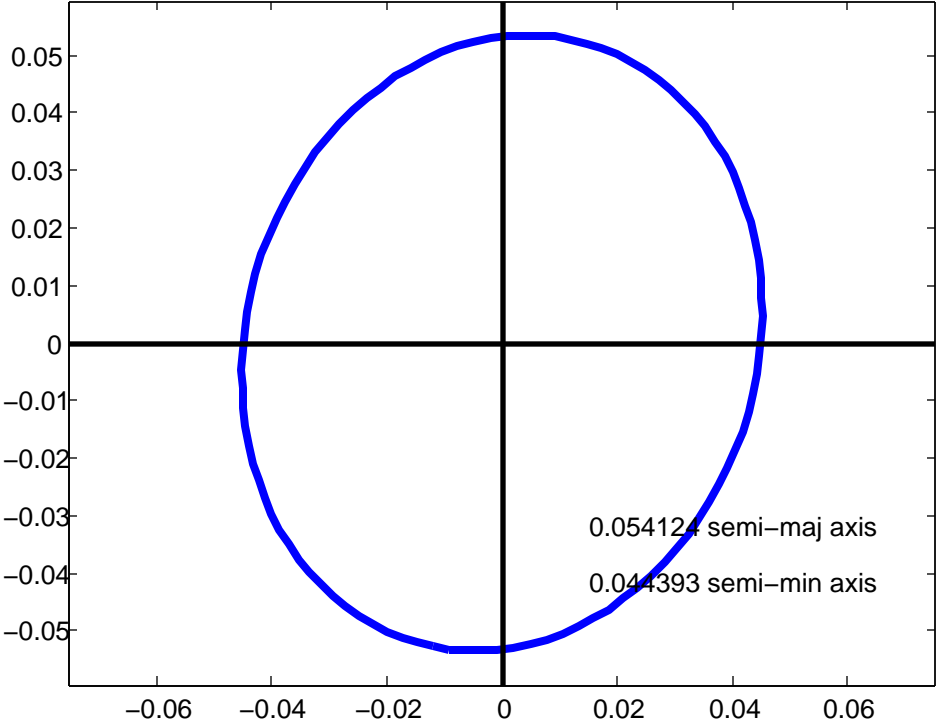
rdt =
    0.121608811495527
residuals (km)
v =
    0.015976052560702
   -0.00779922601096748
    0.0323264025303067
    0.0197165570329339
    0.00165072544625723
   -0.0318763011472966
   -0.024377520539306
   -0.00561668987262968
rms =
    0.0205986507011942
r =
    4
test_stat =
    5.43109645708555
cv1 =
    0.48441855708793
cv2 =
    11.1432867818778
pass global test at alpha = 0.05
convert XYZ to phi,lam,h
phioo =
    0.706126464069501
N =
    6387.14518884767
new_phi =
    0.706126341195602
phioo =
    0.706126341195602
phi
result =
    40
    27
    29.0129526818379
lambda
result =
    -86
    -55
    -33.160920126619
h
h =
    0.236789412511826
J =
    0.99856099354595    0.0536278115209362    0
   -0.034798617800479    0.647957494057097    0.760881161610946
    0.0408043915247028   -0.759786248708623    0.648891252757792
cov enu
mx =
    0.000443302769107075    5.44218791494313e-005
    5.44218791494312e-005    0.000620746424128162
alen =
    0.0541237883266633
blen =
    0.0443930530407871
theta =
   -1.84590307002344
zfact =
    1.64485362695147
half_width =

```

gps_nav51a

width = 0.0930159669289425
diary off 0.186031933857885

90% confidence ellipse for e,n – pass global test



gps_nav51a

```

% gps_nav51a.m 15-nov-11
% solve gps pseudorange problem for 1 epoch
% include elevation angle mask

% note you may need to edit the obs file to replace spaces with
% zeros in satellite number string, and you may need to edit the
% data fields to replace spaces with zeros ??? why are
% spaces there ???

fido=fopen('epoch1.txt','rt');

% interpret satellite prn's in the observation file
S=textscan(fido,'%d %d %d %d %d %f %d %s',1);
% need to edit leading zero into single digit sat number fields
str=char(S{8});
ck1=double(str(1));
ck2=double(str(2));
start_char=0;
proceed=0;
if((ck1 >= 48) & (ck1 <= 57))
    start_char=2;
end
if((ck2 >= 48) & (ck2 <= 57))
    start_char=3;
end
switch start_char
case 0
    disp('cannot interpret satellite string');
    proceed=0;
case 2
    proceed=1;
    numsat=str2num(str(1));
case 3
    proceed=1;
    numsat=str2num(str(1:2));
    if(numsat > 15)
        disp('too many satellites');
        proceed=0;
    end
end
sats=zeros(numsat,1);
if(proceed == 1)
    run_char=start_char;
    for i=1:numsat
        sats(i)=str2num(str(run_char+1:run_char+2));
        run_char=run_char+3;
    end
end

numsat
sats

%disp('pause, press a key to continue');
%pause

% textscan returns cell arrays which are columns from the file
% for 2011 data we want the 3rd column which is C1
% 2011 file has 5 observations (was 6 before eliminating one)
S=textscan(fido,'%f %f %f %f %f',numsat);
c1=S{3};
fclose(fido);
%disp('have read the satellite file');

```

```

%pause

fids=fopen('epoch1s.txt','rt');
% interpret first line of satellite file
S=textscan(fids,'%s %d %d %d %d %d %f',1);
% interpret the satellite data
S=textscan(fids,'%s %f %f %f %f %f',31);
fclose(fids);
tsat=S{2};
XX=S{3};
YY=S{4};
ZZ=S{5};
tdt=S{6};
Xs=zeros(numsat,1);
Ys=zeros(numsat,1);
Zs=zeros(numsat,1);
dt=zeros(numsat,1);
for i=1:numsat
    % last year this number was 31, this year it is 30
    % seems satellite #25 missing
    % now (2011) it is 32, why not just use variable "numsat" ??
    % now (fall, 2011) satellite #24 is missing, so we are back to 31
    for j=1:31
        if(tsat(j) == sats(i))
            Xs(i)=XX(j);
            Ys(i)=YY(j);
            Zs(i)=ZZ(j);
            dt(i)=tdt(j);
        end
    end
end
[m,n]=size(Xs);
m
numsat
%disp('# matches, # from observations');
%pause

if(numsat ~= m)
    disp('error in satellite counting');
    pause
end

% edit out any sats with invalid dt and also corresponding obs
%[c1 Xs Ys Zs dt]
keep_going=1;
while (keep_going==1)
    change=0;
    for i=1:numsat
        if(dt(i) > 999999.0)
            c1(i)=[];
            Xs(i)=[];
            Ys(i)=[];
            Zs(i)=[];
            dt(i)=[];
            change=1;
            break
        end
    end
    if(change == 1)
        numsat=numsat-1;
    else
        keep_going=0;
    end
end

```

```

    end
end

disp('after editing out defective sats');
numsat
disp('c1 xs ys zs dt');
[c1 Xs Ys Zs dt]
%pause

nobs=numsat;

%disp('ok check variables');
%pause

npar=4;
n=nobs;
n0=npar;
r=n-n0;
% Xs,Ys,Zs (km), c1 (m), DT(us)
% we solve in km & us

rawpr=c1/1000; % convert to km
DT=dt;
W=eye(nobs);
sigma0=0.025;

% units
% c1 (unrefined pseudorange) is in meters in file
% xs,xs,zs in km
% DT 1e-06 sec, i.e. us or microseconds
c=0.299792458; % km/us (km / u-second)
pr=rawpr + c*DT;
% initial approximations to receiver coords
% found in the observation file header
% convert from given m to km
Xo= 260604.0/1000;
Yo= -4853018.0/1000;
Zo= 4116926.0/1000;
%Xo=0.0;
%Yo=0.0;
%Zo=0.0;
rdt=0.0;

% compute matrices for the XYZ -> enu conversion
% XYZ -> phi,lambda,h
% note geo2xyz.m and xyz2geo.m require METERS not KM !!
[P,L,h]=xyz2geo(Xo*1000,Yo*1000,Zo*1000);
ax=pi/2-P;
Mx=[1 0 0;0 cos(ax) sin(ax);0 -sin(ax) cos(ax)];
az=L + pi/2;
Mz=[cos(az) sin(az) 0;-sin(az) cos(az) 0;0 0 1];
M=Mx*Mz;

old_phi=9.99e+09;
threshold=1.0e-06;
converged=0;
elev_thresh=10.0;
for iter=1:10
    n_actual_obs=0;
    B=zeros(nobs,npar);
    f=zeros(nobs,1);
    for i=1:nobs

```

```

                                gps_nav51a
enu=M*[Xs(i)-Xo; Ys(i)-Yo; Zs(i)-Zo];
elev=atan(enu(3)/(sqrt(enu(1)^2 + enu(2)^2)));
elevd=57.29577951*elev;
if(iter == 1)
    disp('# sat. id');
    [i sats(i)]
    disp('east, north, up of satellite for elev. angle check');
    enu'
    elevd
    end
if(elevd > elev_thresh)
    % only use this observations if above elevation angle mask
    % otherwise ignore it
    % increment the actual observation count
    n_actual_obs=n_actual_obs + 1;
    D=sqrt((Xs(i)-Xo)^2 + (Ys(i)-Yo)^2 + (Zs(i)-Zo)^2);
    B(i,1)=(Xs(i)-Xo)/D;
    B(i,2)=(Ys(i)-Yo)/D;
    B(i,3)=(Zs(i)-Zo)/D;
    B(i,4)=-c;
    F=pr(i) - D - c*rdt;
    f(i)=-F;
else
    disp('reject low satellite number');
    i
    end
end
%B
%f
%pause
% W
%condJ=cond(B)
N=B'*W*B;
condN=cond(N)
t=B'*W*f;

%keyboard

Ni=inv(N);
del=Ni*t;
Qdd=Ni;
Xo=Xo + del(1);
Yo=Yo + del(2);
Zo=Zo + del(3);
rdt=rdt + del(4);
disp_del=[iter; del(1); del(2); del(3); del(4)]
v=f-B*del;
phi=v'*W*v;
if(abs(phi-old_phi)/phi < threshold)
    disp('we have converged');
    converged=1;
    break;
end

old_phi=phi;
end

if(converged == 0)
    disp('we did not converge');
else
    % we converged
    % show results

```

```

                                gps_nav51a
disp('receiver location (km)');
[Xo Yo Zo]
disp('receiver clock bias (usec)');
[rdt]
end
disp('residuals (km)');
v
rms=sqrt(v'*v/nobs)

% make global test and post adjustment statistics
% careful - you must modify redundancy if you reject satellites
% low on the horizon

r
test_stat=v'*W*v/(sigma0^2)
alpha = 0.05;
cv1=icdf('chi2',alpha/2,r)
cv2=icdf('chi2',1-alpha/2,r)
if((test_stat > cv1) && (test_stat < cv2))
    Sdd=sigma0^2 * Qdd;
    pass=1;
    disp('pass global test at alpha = 0.05');
else
    sigma0_hat_sqr=v'*W*v/r;
    Sdd=sigma0_hat_sqr * Qdd;
    pass=0;
    disp('fail global test at alpha = 0.05');
end

% ok transform cofactor matrix to ENU from XYZ
% first get lat, lon, h from XYZ by iteration
% remember Xo,Yo,Zo are km

disp('convert XYZ to phi,lam,h');
lam=atan2(Yo,Xo);
a=6378137.0/1000;
f=1/298.257223563;
esqr=2*f-f^2;
e=sqrt(esqr);
phioo=atan(Zo/((1-e^2)*sqrt(Xo^2+Yo^2)))
keep_going=1;
while(keep_going == 1)
    N=a/sqrt(1-esqr*(sin(phioo))^2)
    new_phi=atan((Zo/sqrt(Xo^2 + Yo^2))*(1 + esqr*N*sin(phioo)/Zo))
    if(abs(new_phi - phioo) < 1.0e-06);
        keep_going=0;
    end
    phioo=new_phi
end
N=a/sqrt(1-esqr*(sin(phioo))^2);
phi=phioo;
h=sqrt(Xo^2 + Yo^2)/cos(phi) - N;
disp('phi');
result=raddms(phi)
disp('lambda');
result=raddms(lam)
disp('h');
h

% rotate XYZ into enu
% bvg uses "enu" for local cartesian and ENH for map projections
% extract the submatrix for XYZ to transform

```

```

M=m1(pi/2 - phi)*m3(lam + pi/2);
J=M;
J
Sdd_enu=J*Sdd(1:3,1:3)*J';
disp('cov enu');
mx=Sdd_enu(1:2,1:2)
sigz_sqr=Sdd_enu(3,3);
[V,D]=eig(mx);
if(D(1,1) < D(2,2))
    tmp=D(1,1);
    D(1,1)=D(2,2);
    D(2,2)=tmp;
    tmp=V(:,1);
    V(:,1)=V(:,2);
    V(:,2)=tmp;
end

% 90%
P=0.90;
alpha=1-P;
if(pass == 1)
    alen=sqrt(D(1,1)*icdf('chi2',P,2))
    blen=sqrt(D(2,2)*icdf('chi2',P,2))
    theta=atan2(V(2,1),V(1,1))
    plot_ell(0,0,alen,blen,theta);
    title('90% confidence ellipse for e,n - pass global test');
    zfact=icdf('norm',1-alpha/2,0,1)
    half_width=zfact*sqrt(sigz_sqr)
    width=2*half_width
else
    alen=sqrt(D(1,1)*2*icdf('f',P,2,r))
    blen=sqrt(D(2,2)*2*icdf('f',P,2,r))
    theta=atan2(V(2,1),V(1,1))
    plot_ell(0,0,alen,blen,theta);
    title('90% confidence ellipse for e,n - fail global test');
    zfact=icdf('t',1-alpha/2,r)
    half_width=zfact*sqrt(sigz_sqr)
    width=2*half_width
end
axis equal

```

gps_nav53a

gps_nav53a

numsat =

8

sats =

16

13

23

10

6

7

19

3

numsat =

8

sats =

16

13

23

10

6

7

19

3

numsat =

9

sats =

16

13

11

23

28

6

7

19

3

nobs =

25

found no observations to throw out
after editing out defective sats

nobs =

25

display c1 xs ys zs dt

ans =

Columns 1 through 3

21287141.34	9671.17665	-11273.286067
20287554.48	-5734.807399	-20258.741892
20605103.28	-1126.552213	-25140.371937
24293518.04	-22017.451052	-5852.11655
21364948.06	14140.406396	-14804.666272
22499418.9	-15186.548693	-8878.439153
21662186.34	8347.789114	-24844.425342
20253866.26	11027.953133	-19286.137311
22105196.56	13581.808579	-8214.458315
20476109.34	-4857.487454	-23346.182827
21408621.52	-450.357415	-26256.949516
24797214.06	-23866.148357	-7811.771947
21822514.44	14311.668857	-10596.944933
21648563.4	-10869.754054	-10898.301998
20909935.44	8669.138898	-23462.439265
20365537.74	11663.989807	-15502.864843
23021211.5	17521.44413	-5769.712404
21076358.28	-4401.494092	-25383.444655

gps_nav53a

24209534.96	10059.980919	-23028.026064
22497845.42	118.087873	-26137.930493
24951548.94	-23182.08437	-13171.392156
22605053.64	14916.897496	-5946.74466
20920105.78	-6602.066393	-13489.883887
20542044.8	9046.355443	-20922.243207
20897918.8	12741.361601	-11095.36145

Columns 4 through 5

22027.954669	-200.884996
16005.394146	248.828766
7915.839784	271.527222
13742.241308	-14.912353
16924.338317	-58.414824
19985.16409	29.497196
2801.519349	-189.641841
13886.297212	775.502939
21357.257239	-200.888751
11474.898571	248.8252
2331.17586	271.521772
8997.979116	-14.913646
19770.511792	-58.4
21684.257488	29.500961
8340.227951	-189.649224
17563.988766	775.511615
19230.899452	-200.892787
6149.175722	248.822033
-8412.018346	-190.787419
-3416.760203	271.516024
1904.008804	71.968727
21269.077697	-58.38542
21902.536054	29.50463
13294.801402	-189.656602
19983.866332	775.521111

press key to continue

epoch sat. id

ans =

1 1 16

enu sat coords, elev ang

ans =

9052.76533048471 9140.70388902187 16884.2719688828

elevd =

52.694616383131

epoch sat. id

ans =

2 1 13

enu sat coords, elev ang

ans =

-6812.87323234033 -727.888551905959 19174.7724344859

elevd =

70.336864806318

epoch sat. id

ans =

3 1 23

enu sat coords, elev ang

ans =

-2473.01125170978 -10206.5150230189 17822.5684034722

elevd =

59.4915299610946

epoch sat. id

ans =

4 1 10

enu sat coords, elev ang


```

gps_nav53a
ans =
    -22299.577380379      7451.52569989327      6095.92019233292
elevd =
    14.5351729699529
# epoch sat. id
ans =
    5      1      6
enu sat coords, elev ang
ans =
    13326.2051335967      2813.76579120993      16438.0353737534
elevd =
    50.3559565824672
# epoch sat. id
ans =
    6      1      7
enu sat coords, elev ang
ans =
    -15640.7804486962      10003.0547395423      12724.9520507625
elevd =
    34.4264003216796
# epoch sat. id
ans =
    7      1      19
enu sat coords, elev ang
ans =
    7003.56861856772      -14235.8109717316      14665.6379957201
elevd =
    42.749785202393
# epoch sat. id
ans =
    8      1      3
enu sat coords, elev ang
ans =
    9977.9238214494      -2293.3252966074      17744.6625657583
elevd =
    60.0164812847079
# epoch sat. id
ans =
    9      2      16
enu sat coords, elev ang
ans =
    13121.7918327613      10476.2972687038      14284.5569862263
elevd =
    40.3888833650843
# epoch sat. id
ans =
    10     2     13
enu sat coords, elev ang
ans =
    -6102.37058774596      -6206.11604978534      18616.5776283423
elevd =
    64.9427153072618
# epoch sat. id
ans =
    11     2     23
enu sat coords, elev ang
ans =
    -1857.66254204288      -15202.8105716079      15074.6905321948
elevd =
    44.5452806306833
# epoch sat. id
ans =

```

gps_nav53a

```

12      2      10
enu sat coords, elev ang
ans =
      -24250.6958228289          2636.25333472451          4430.91460083473
elevd =
      10.2951165990729
# epoch sat. id
ans =
13      2      6
enu sat coords, elev ang
ans =
      13722.8480991247          7699.82859420493          15094.8980799137
elevd =
      43.8097827842802
# epoch sat. id
ans =
14      2      7
enu sat coords, elev ang
ans =
      -11438.5056927837          9836.88167294048          15538.2692441828
elevd =
      45.8450785992012
# epoch sat. id
ans =
15      2      19
enu sat coords, elev ang
ans =
      7398.56108708732          -9137.21843862679          17222.7416912608
elevd =
      55.6807977273912
# epoch sat. id
ans =
16      2      3
enu sat coords, elev ang
ans =
      10815.912484779          2934.22943639479          17282.5425621228
elevd =
      57.0386237136445
# epoch sat. id
ans =
17      3      16
enu sat coords, elev ang
ans =
      17186.8518649953          10305.3966936034          11208.0348897423
elevd =
      29.2179314449479
# epoch sat. id
ans =
18      3      13
enu sat coords, elev ang
ans =
      -5756.27554302217          -11594.2886384276          16727.2645411605
elevd =
      52.2650975877687
# epoch sat. id
ans =
19      3      11
enu sat coords, elev ang
ans =
      8810.69625603638          -21650.6274970878          6079.07172341916
elevd =
      14.577985478495

```

gps_nav53a

```

# epoch sat. id
ans =
  20      3      23
enu sat coords, elev ang
ans =
  -1283.65302521722      -19518.9759524361      11277.6793342188
elevd =
  29.9649488368088
# epoch sat. id
ans =
  21      3      28
enu sat coords, elev ang
ans =
  -23855.0101476307      -6258.0253769785      3927.79815749533
elevd =
  9.04914746594903
reject satellite too low, #
i =
  21
# epoch sat. id
ans =
  22      3      6
enu sat coords, elev ang
ans =
  14576.5595498052      11832.1270431225      12558.8282180696
elevd =
  33.7800012499309
# epoch sat. id
ans =
  23      3      7
enu sat coords, elev ang
ans =
  -7315.92404657428      8175.24210964246      17823.083180511
elevd =
  58.3862082929748
# epoch sat. id
ans =
  24      3      19
enu sat coords, elev ang
ans =
  7911.44560501917      -3734.55926023216      18523.0915248167
elevd =
  64.7182843111287
# epoch sat. id
ans =
  25      3      3
enu sat coords, elev ang
ans =
  12128.0738989628      7593.85306249333      15547.9637479794
elevd =
  47.3755708130742
disp_del =
  1
  0.0350637592224287
  -0.00421682058919907
  0.0111656415605083
disp_del2 =
  0.0277018313953836      -0.000245000957323688      0.00626409559606281
reject satellite too low, #
i =
  21
disp_del =

```

gps_nav53a

```

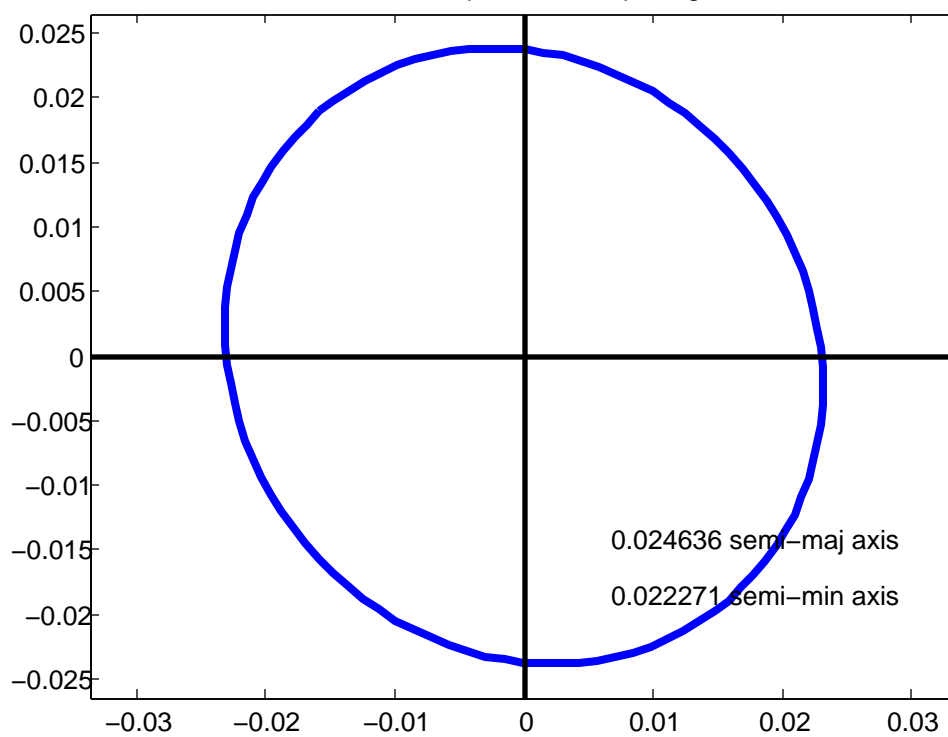
                2
-9.43920348229025e-009
-1.81159965780628e-008
 3.05933441911158e-008
disp_del2 =
 9.08167720975492e-009      1.00483116226525e-008      -1.12043524979485e-008
reject satellite too low, #
i =
 21
disp_del =
                3
 4.6118886124417e-013
-2.56353202256107e-012
 1.96981333253682e-012
disp_del2 =
 7.78562323968246e-012      8.34202517301392e-012      2.44768091340789e-012
we have converged
receiver location (km)
ans =
    260.639063749784      -4853.02221683871      4116.93716567216
receiver clock biases (usec)
ans =
    0.0277018404848465      -0.00024499090067004      0.00626408439415799
residuals (km) & elevation angle (deg)
ans =
    0.020685495191302      52.6946564362566
    0.00275504279197603      70.3367606233464
    0.0331140763891361      59.4914831722901
    0.0112591013853823      14.5351271626833
    0.000630164188899284      50.3560085234864
    -0.0280950963439292      34.4263367252924
    -0.0357002160295635      42.7497823726633
    -0.0046485675732031      60.0165443449799
    0.0151696439170647      40.388910405275
    0.00765944618913502      64.9426314973891
    0.0355388545393485      44.5452417314939
    0.0106801747517191      10.2950755062754
    0.00391418475905088      43.809819487134
    -0.0347167610050918      45.845012266326
    -0.0357868924408598      55.6808209884346
    -0.00245865071036665      57.0386868679011
    0.0175531045087651      29.2179447590705
    0.0231834534770693      52.2650360338658
    -0.0572258834366306      14.5779635202827
    0.0457940440673544      29.9649140815365
    0.0161309556231242      33.7800183340489
    -0.0305736770405597      58.3861450547179
    -0.0231737870003826      64.7183447756052
    0.00831178980126003      47.3756126360454
rms =
    0.0257684933192967
elevation threshold and rejections
ans =
 10      1
new number of observations
nobs =
 24
r =
 18
test_stat =
 25.4981855211509
cv1 =
```

```

      8.23074619475666
cv2 =      31.5263784403866
pass global test
convert XYZ to phi,lam,h
phioo =      0.706124683635765
N =      6387.14515115495
new_phi =      0.706124581013248
phioo =      0.706124581013248
phi
result =
                40
                27
      28.649889009809
lambda
result =
                -86
                -55
      -32.862061950857
h
h =      0.197762023461109
J =
      0.99856091584328      0.0536292583412191      0
      -0.0347994848043434      0.647956106273267      0.7608823037767
      0.0408055536365026      -0.759787330108207      0.648889913467193
alen =      0.0246356432552529
blen =      0.0222712398560352
theta =      2.21036849452655
zfact =      1.64485362695147
half_width =      0.0428091983363348
width =      0.0856183966726697
diary off

```

90% confidence ellipse for e,n – pass global test



gps_nav53a

```

% gpsnav53a.m 26-nov-2011
% this will be the 3-epoch solution

% 3 epochs extracted from NGS/CORS RINEX files
obsfile=['epoch1.txt'; 'epoch2.txt'; 'epoch3.txt'];
satfile=['epoch1s.txt'; 'epoch2s.txt'; 'epoch3s.txt'];
nobs=0;

for k=1:3
    % maybe need an epoch = 1,2,3,4 variable for multiple epoch case
    fido=fopen(obsfile(k,:), 'rt');

    % interpret satellite prn's in the observation file
    S=textscan(fido,'%d %d %d %d %d %f %d %s',1);
    str=char(S{8});
    ck1=double(str(1));
    ck2=double(str(2));
    start_char=0;
    proceed=0;
    if((ck1 >= 48) & (ck1 <= 57))
        start_char=2;
    end
    if((ck2 >= 48) & (ck2 <= 57))
        start_char=3;
    end
    switch start_char
        case 0
            disp('cannot interpret satellite string');
            proceed=0;
        case 2
            proceed=1;
            numsat=str2num(str(1));
        case 3
            proceed=1;
            numsat=str2num(str(1:2));
            if(numsat > 15)
                disp('too many satellites');
                proceed=0;
            end
        end
    end
    sats=zeros(numsat,1);
    if(proceed == 1)
        run_char=start_char;
        for i=1:numsat
            sats(i)=str2num(str(run_char+1:run_char+2));
            run_char=run_char+3;
        end
    end

    numsat
    sats

    % multi-epoch satellite ID
    if(k==1)
        totsats=sats;
    else
        totsats=[totsats;sats];
    end

    %disp('pause, press a key to continue');
    %pause

```

```

                                gps_nav53a
S=textscan(fido,'%f %f %f %f %f',numsat);
tcl=S{3};
fclose(fido);

fids=fopen(satfile(k,:), 'rt');
% interpret first line of satellite file
S=textscan(fids,'%s %d %d %d %d %d %f',1);
% interpret the satellite data
S=textscan(fids,'%s %f %f %f %f %f',31);
fclose(fids);
tsat=S{2};
XX=S{3};
YY=S{4};
ZZ=S{5};
TT=S{6};
tXs=zeros(numsat,1);
tYs=zeros(numsat,1);
tZs=zeros(numsat,1);
tdt=zeros(numsat,1);
for i=1:numsat
    % last year this number was 31, this year it is 30
    % seems satellite #25 missing
    % 2011 seems to be 32 this year
    % fall 2011 sat #24 missing back to 31
    for j=1:31
        if(tsat(j) == sats(i))
            tXs(i)=XX(j);
            tYs(i)=YY(j);
            tZs(i)=ZZ(j);
            tdt(i)=TT(j);
        end
    end
end
[m,n]=size(tXs);
if(numsat ~= m)
    disp('error in satellite counting');
    pause
end
nobs=nobs+numsat;
tepch=ones(numsat,1);
tepch=tepch*k;
% transfer data into big arrays
% k is epoch counter 1,2,3, .....
% variable "epoch" will look like: [1;1;1;1;1;2;2;2;2;2;2;3;3;3;3;3]
% i.e. an epoch attribute for each observation
if(k == 1)
    epoch=tepch;
    c1=tcl;
    Xs=tXs;
    Ys=tYs;
    Zs=tZs;
    dt=tdt;
else
    epoch=[epoch; tepch];
    c1=[c1; tcl];
    Xs=[Xs; tXs];
    Ys=[Ys; tYs];
    Zs=[Zs; tZs];
    dt=[dt; tdt];
end
end

```



```

nobs

% edit out any sats with invalid dt and also corresponding obs
%[c1 Xs Ys Zs dt]
keep_going=1;
while (keep_going==1)
    change=0;
    for i=1:nobs
        if(dt(i) > 999999.0)
            c1(i)=[];
            Xs(i)=[];
            Ys(i)=[];
            Zs(i)=[];
            dt(i)=[];
            epoch(i)=[];
            totsats(i)=[];
            change=1;
            break
        end
    end
    if(change == 1)
        nobs=nobs-1
        disp('threw one out');
    else
        keep_going=0;
        disp('found no observations to throw out');
    end
end

disp('after editing out defective sats');
nobs
disp('display c1 xs ys zs dt');
[c1 Xs Ys Zs dt]
disp('press key to continue');
pause

%disp('ok check variables');
%pause

npar=3 + 3;
%n=nobs;
%n0=npar;
%r=n-n0;
% Xs,Ys,Zs (km), c1 (m), DT(us)
% we solve in km & us
rawpr=c1/1000; % convert to km
DT=dt;
%
sigma0=0.025;

% units
% c1 (unrefined pseudorange) is in meters in file
% xs,xs,zs in km
% DT 1e-06 sec, i.e. us or microseconds
c=0.299792458; % km/us (km / u-second)
pr=rawpr + c*DT;
% initial approximations to receiver coords
% found in the observation file header
% convert from given m to km
% these are INWL coords
Xo= 260604.0/1000;
Yo= -4853018.0/1000;

```

gps_nav53a

```

Zo= 4116926.0/1000;
%Xo=0.0;
%Yo=0.0;
%Zo=0.0;
% fixed station coords for comparison (km)
Xfx=260604.165/1000;
Yfx=-4853018.928/1000;
Zfx=4116926.538/1000;

% compute matrices for the XYZ -> enu conversion
% XYZ -> phi,lambda,h
% note geo2xyz.m and xyz2geo.m require METERS not KM !!
[P,L,h]=xyz2geo(Xo*1000,Yo*1000,Zo*1000);
ax=pi/2-P;
Mx=[1 0 0;0 cos(ax) sin(ax);0 -sin(ax) cos(ax)];
az=L + pi/2;
Mz=[cos(az) sin(az) 0;-sin(az) cos(az) 0;0 0 1];
M=Mx*Mz;

% test M see if correct
% note geo2xyz.m and xyz2geo.m require METERS not KM !!

rdt=[0.0; 0.0; 0.0];
old_phi=9.99e+09;
threshold=1.0e-09;
converged=0;
elev_thresh=10.0; % degrees
for iter=1:10
    ii=0; % counter for obs passing elev threshold
    B=zeros(nobs,npar);
    f=zeros(nobs,1);
    elev_disp=zeros(nobs,1);
    for i=1:nobs
        enu=M*[Xs(i)-Xo; Ys(i)-Yo; Zs(i)-Zo];
        elev=atan(enu(3)/(sqrt(enu(1)^2 + enu(2)^2)));
        elevd=57.29577951*elev;
        if(iter == 1)
            disp('# epoch sat. id');
            [i epoch(i) totsats(i)]
            disp('enu sat coords, elev ang');
            enu'
            elevd
        end
        if(elevd > elev_thresh)
            % only use this observations if above elevation angle mask
            % otherwise ignore it
            % ii counts actually used observations
            ii=ii+1;
            elev_disp(ii)=elevd;
            D=sqrt((Xs(i)-Xo)^2 + (Ys(i)-Yo)^2 + (Zs(i)-Zo)^2);
            B(ii,1)=(Xs(i)-Xo)/D;
            B(ii,2)=(Ys(i)-Yo)/D;
            B(ii,3)=(Zs(i)-Zo)/D;
            idx=3+epoch(i);
            B(ii,idx)=-c;
            F=pr(i) - D - c*rdt(epoch(i));
            f(ii)=-F;
        else
            disp('reject satellite too low, #');
            i
        end
    end
end

```

```

Btemp=B(1:ii,:);
ftemp=f(1:ii);
etemp=elev_disp(1:ii);
B=Btemp;
f=ftemp;
elev_disp=etemp;
rej=nobs-ii;
nobs_new=ii;
W=eye(nobs_new);
% B
% f
% W
%condJ=cond(B)
N=B'*W*B;
%condN=cond(N)
t=B'*W*f;
Ni=inv(N);
del=Ni*t;
Qdd=Ni;
Xo=Xo + del(1);
Yo=Yo + del(2);
Zo=Zo + del(3);
rdt(1)=rdt(1) + del(4);
rdt(2)=rdt(2) + del(5);
rdt(3)=rdt(3) + del(6);
disp_del=[iter; del(1); del(2); del(3)]
disp_del2=[del(4) del(5) del(6)]
v=f-B*del;
phi=v'*W*v;
if(abs(phi-old_phi)/phi < threshold)
    disp('we have converged');
    converged=1;
    break;
end

old_phi=phi;
end

disp('enu displacements from known fixed station coords');
denu=M*[Xo-Xfx;Yo-Yfx;Zo-Zfx]

nobs=nobs_new;
n=nobs;
n0=npar;
r=n-n0;

if(converged == 0)
    disp('we did not converge');
else
    % we converged
    % show results
    disp('receiver location (km)');
    [Xo Yo Zo]
    disp('receiver clock biases (usec)');
    [rdt]
end
disp('residuals (km) & elevation angle (deg)');
[v elev_disp]
rms=sqrt(v'*v/nobs)
disp('elevation threshold and rejections');
[elev_thresh rej]
disp('new number of observations');

```

```

nobs

% make global test and post adjustment statistics

r
alpha=0.05;
test_stat=v'*W*v/(sigma0^2)
cv1=icdf('chi2',alpha/2,r)
cv2=icdf('chi2',1-alpha/2,r)
if((test_stat > cv1) && (test_stat < cv2))
    Sdd=sigma0^2 * Qdd;
    pass=1;
    disp('pass global test');
else
    sigma0_hat_sqr=v'*W*v/r;
    Sdd=sigma0_hat_sqr * Qdd;
    pass=0;
    disp('fail global test');
end

% ok transform cofactor matrix to ENU from XYZ
% first get lat, lon, h from XYZ by iteration
% remember Xo,Yo,Zo are km

disp('convert XYZ to phi,lam,h');
lam=atan2(Yo,Xo);
a=6378137.0/1000;
f=1/298.257223563;
esqr=2*f-f^2;
e=sqrt(esqr);
phioo=atan(Zo/((1-e^2)*sqrt(Xo^2+Yo^2)))
keep_going=1;
while(keep_going == 1)
    N=a/sqrt(1-esqr*(sin(phioo))^2)
    new_phi=atan((Zo/sqrt(Xo^2 + Yo^2))* (1 + esqr*N*sin(phioo)/Zo))
    if(abs(new_phi - phioo) < 1.0e-06);
        keep_going=0;
    end
    phioo=new_phi
    %pause
end
N=a/sqrt(1-esqr*(sin(phioo))^2);
phi=phioo;
h=sqrt(Xo^2 + Yo^2)/cos(phi) - N;
disp('phi');
result=raddms(phi)
disp('lambda');
result=raddms(lam)
disp('h');
h

% rotate XYZ into ENU
% extract the submatrix for XYZ to transform
M=m1(pi/2 - phi)*m3(lam + pi/2);
J=M;
J
Sdd_enu=J*Sdd(1:3,1:3)*J';
mx=Sdd_enu(1:2,1:2);
sigz_sqr=Sdd_enu(3,3);
[V,D]=eig(mx);
if(D(1,1) < D(2,2))
    tmp=D(1,1);

```

```

D(1,1)=D(2,2);
D(2,2)=tmp;
tmp=V(:,1);
V(:,1)=V(:,2);
V(:,2)=tmp;
end

% 90%
P=0.90;
alpha=1-P;
if(pass == 1)
    alen=sqrt(D(1,1)*icdf('chi2',P,2))
    blen=sqrt(D(2,2)*icdf('chi2',P,2))
    theta=atan2(V(2,1),V(1,1))
    plot_ell(0,0,alen,blen,theta);
    title('90% confidence ellipse for e,n - pass global test');
    zfact=icdf('norm',1-alpha/2,0,1)
    half_width=zfact*sqrt(sigz_sqr)
    width=2*half_width
else
    alen=sqrt(D(1,1)*2*icdf('f',P,2,r))
    blen=sqrt(D(2,2)*2*icdf('f',P,2,r))
    theta=atan2(V(2,1),V(1,1))
    plot_ell(0,0,alen,blen,theta);
    title('90% confidence ellipse for e,n - fail global test');
    zfact=icdf('t',1-alpha/2,r)
    half_width=zfact*sqrt(sigz_sqr)
    width=2*half_width
end
axis equal

```

gps_nav51b

gps_nav51b

numsat =

8

sats =

16

13

23

10

6

7

19

3

m =

8

numsat =

8

after editing out defective sats

numsat =

8

ans =

Columns 1 through 3

21287141.34	9671.17665	-11273.286067
20287554.48	-5734.807399	-20258.741892
20605103.28	-1126.552213	-25140.371937
24293518.04	-22017.451052	-5852.11655
21364948.06	14140.406396	-14804.666272
22499418.9	-15186.548693	-8878.439153
21662186.34	8347.789114	-24844.425342
20253866.26	11027.953133	-19286.137311

Columns 4 through 5

22027.954669	-200.884996
16005.394146	248.828766
7915.839784	271.527222
13742.241308	-14.912353
16924.338317	-58.414824
19985.16409	29.497196
2801.519349	-189.641841
13886.297212	775.502939

c1 xs ys zs dt

i =

1

ans =

9052.76533048471	9140.70388902187	16884.2719688828
------------------	------------------	------------------

elevd =

52.694616383131

i =

2

ans =

-6812.87323234033	-727.888551905959	19174.7724344859
-------------------	-------------------	------------------

elevd =

70.336864806318

i =

3

ans =

-2473.01125170978	-10206.5150230189	17822.5684034722
-------------------	-------------------	------------------

elevd =

59.4915299610946

i =

4

ans =

-22299.577380379	7451.52569989327	6095.92019233292
------------------	------------------	------------------

elevd =

gps_nav51b

```

14.5351729699529
i =
  5
ans =
13326.2051335967      2813.76579120993      16438.0353737534
elevd =
50.3559565824672
i =
  6
ans =
-15640.7804486962      10003.0547395423      12724.9520507625
elevd =
34.4264003216796
i =
  7
ans =
7003.56861856772      -14235.8109717316      14665.6379957201
elevd =
42.749785202393
i =
  8
ans =
9977.9238214494      -2293.3252966074      17744.6625657583
elevd =
60.0164812847079
g =
-0.000237229328675009
disp_del =
1
0.010695376270385
-0.0588965894481257
0.0449588094882004
0.179254604835048
g =
0.0249177233026189
disp_del =
2
0.0146137952969054
0.0203558628875574
0.00853413767083341
-0.0235287330630057
g =
0.000398519601232453
disp_del =
3
0.0002132990297104
0.000367967769449774
0.000260766036245701
-0.000260407228760294
g =
4.50388145625813e-007
disp_del =
4
2.24555731739161e-006
1.07865011263511e-006
1.63274677696885e-005
2.34504948483273e-005
we have converged
receiver location (km)
ans =
260.629524716154      -4853.05617168014      4116.97977004066
receiver clock bias (usec)

```

```
rdt =
      0.15548891503813
residuals (km)
v =
      0.0170051076080605
     -0.0123083961524231
      0.0296604915340265
      0.0225394704410416
      0.00375251685268497
     -0.0324405149146357
     -0.0226445037116168
     -0.00556417165713793
rms =
      0.0207323561481844
constraint misclosure (meters)
constr_misc_m =
     -1.36324615485073e-006
r =
      5
test_stat =
      5.50183157062602
alpha =
      0.05
cv1 =
      0.831211613486658
cv2 =
      12.83250199403
pass global test at alpha = 0.05
diary off
```



```

                                gps_nav51b
% gps_nav51b.m 15-nov-11
% solve gps pseudorange problem for 1 epoch
% include elevation angle mask
% ok for this one add distance constraint for the receiver position

% note you may need to edit the obs file to replace spaces with
% zeros in satellite number string, and you may need to edit the
% data fields to replace spaces with zeros ??? why are
% spaces there ???

fido=fopen('epoch1.txt','rt');

% interpret satellite prn's in the observation file
S=textscan(fido,'%d %d %d %d %d %f %d %s',1);
% need to edit leading zero into single digit sat number fields
str=char(S{8});
ck1=double(str(1));
ck2=double(str(2));
start_char=0;
proceed=0;
if((ck1 >= 48) & (ck1 <= 57))
    start_char=2;
end
if((ck2 >= 48) & (ck2 <= 57))
    start_char=3;
end
switch start_char
case 0
    disp('cannot interpret satellite string');
    proceed=0;
case 2
    proceed=1;
    numsat=str2num(str(1));
case 3
    proceed=1;
    numsat=str2num(str(1:2));
    if(numsat > 15)
        disp('too many satellites');
        proceed=0;
    end
end
sats=zeros(numsat,1);
if(proceed == 1)
    run_char=start_char;
    for i=1:numsat
        sats(i)=str2num(str(run_char+1:run_char+2));
        run_char=run_char+3;
    end
end

numsat
sats

%disp('pause, press a key to continue');
%pause

% textscan returns cell arrays which are columns from the file
% for 2011 data we want the 3rd column which is C1
% 2011 file has 5 observations (was 6 before eliminating one)
S=textscan(fido,'%f %f %f %f %f',numsat);
c1=S{3};
fclose(fido);

```

```

                                gps_nav51b
%disp('have read the satellite file');
%pause

fids=fopen('epoch1s.txt','rt');
% interpret first line of satellite file
S=textscan(fids,'%s %d %d %d %d %d %f',1);
% interpret the satellite data
S=textscan(fids,'%s %f %f %f %f %f',31);
fclose(fids);
tsat=S{2};
XX=S{3};
YY=S{4};
ZZ=S{5};
tdt=S{6};
Xs=zeros(numsat,1);
Ys=zeros(numsat,1);
Zs=zeros(numsat,1);
dt=zeros(numsat,1);
for i=1:numsat
    % last year this number was 31, this year it is 30
    % seems satellite #25 missing
    % now (2011) it is 32, why not just use variable "numsat" ??
    % now (fall, 2011) satellite #24 is missing, so we are back to 31
    for j=1:31
        if(tsat(j) == sats(i))
            Xs(i)=XX(j);
            Ys(i)=YY(j);
            Zs(i)=ZZ(j);
            dt(i)=tdt(j);
        end
    end
end
[m,n]=size(Xs);
m
numsat
%disp('# matches, # from observations');
%pause

if(numsat ~= m)
    disp('error in satellite counting');
    pause
end

% edit out any sats with invalid dt and also corresponding obs
%[c1 Xs Ys Zs dt]
keep_going=1;
while (keep_going==1)
    change=0;
    for i=1:numsat
        if(dt(i) > 999999.0)
            c1(i)=[];
            Xs(i)=[];
            Ys(i)=[];
            Zs(i)=[];
            dt(i)=[];
            change=1;
            break
        end
    end
    if(change == 1)
        numsat=numsat-1;
    else

```

```

        keep_going=0;
        end
    end

disp('after editing out defective sats');
numsat
[c1 Xs Ys Zs dt]
disp('c1 xs ys zs dt');
%pause

nobs=numsat;

%disp('ok check variables');
%pause

npar=4;
n=nobs;
n0=npar-1; % we have a constraint !
r=n-n0;
% Xs,Ys,Zs (km), c1 (m), DT(us)
% we solve in km & us

rawpr=c1/1000; % convert to km
DT=dt;
W=eye(nobs);
sigma0=0.025;

% units
% c1 (unrefined pseudorange) is in meters in file
% xs,ys,zs in km
% DT 1e-06 sec, i.e. us or microseconds
c=0.299792458; % km/us (km / u-second)
pr=rawpr + c*DT;
% initial approximations to receiver coords
% found in the observation file header
% convert from given m to km
Xo= 260604.0/1000;
Yo= -4853018.0/1000;
Zo= 4116926.0/1000;
%Xo=0.0;
%Yo=0.0;
%Zo=0.0;
rdt=0.0;

% compute matrices for the XYZ -> enu conversion
% XYZ -> phi,lambda,h
% note geo2xyz.m and xyz2geo.m require METERS not KM !!
[P,L,h]=xyz2geo(Xo*1000,Yo*1000,Zo*1000);
ax=pi/2-P;
Mx=[1 0 0;0 cos(ax) sin(ax);0 -sin(ax) cos(ax)];
az=L + pi/2;
Mz=[cos(az) sin(az) 0;-sin(az) cos(az) 0;0 0 1];
M=Mx*Mz;

old_phi=9.99e+09;
threshold=1.0e-06;
converged=0;
elev_thresh=10.0;
for iter=1:10
    n_actual_obs=0;
    B=zeros(nobs,npar);
    f=zeros(nobs,1);

```

```

                                gps_nav51b
for i=1:nobs
    enu=M*[Xs(i)-Xo; Ys(i)-Yo; Zs(i)-Zo];
    elev=atan(enu(3)/(sqrt(enu(1)^2 + enu(2)^2)));
    elevd=57.29577951*elev;
    if(iter == 1)
        i
        enu'
        elevd
        end
    if(elevd > elev_thresh)
        % only use this observations if above elevation angle mask
        % otherwise ignore it
        % increment the actual observation count
        n_actual_obs=n_actual_obs + 1;
        D=sqrt((Xs(i)-Xo)^2 + (Ys(i)-Yo)^2 + (Zs(i)-Zo)^2);
        B(i,1)=(Xs(i)-Xo)/D;
        B(i,2)=(Ys(i)-Yo)/D;
        B(i,3)=(Zs(i)-Zo)/D;
        B(i,4)=-c;
        F=pr(i) - D - c*rdt;
        f(i)=-F;
    else
        disp('reject low satellite number');
        i
        end
    end
end
%B
%f
%pause
% W
%condJ=cond(B)

% add constraint equation
% Fc = 100 - [(X-Xf)^2 + (Y-Yf)^2 + (Z-Zf)^2]^(1/2) = 0;
% Xf,Yf,Zf : 260679.818, -4852970.302, 4116969.922 (m)
Xf=260.679818; % (km)
Yf=-4852.970302;
Zf=4116.969922;
D=sqrt((Xo-Xf)^2 + (Yo-Yf)^2 + (Zo-Zf)^2);
% note units here are km !!!
Fc=0.100 - D;
dFcdX=-(Xo-Xf)/D;
dFcdY=-(Yo-Yf)/D;
dFcdZ=-(Zo-Zf)/D;
C=[dFcdX dFcdY dFcdZ 0];
g= -Fc

N=B'*W*B;
%condN=cond(N)
t=B'*W*f;

%keyboard

Nc=[-N C';C 0];
z=[-t; g];
delc=inv(Nc)*z;
Xo=Xo + delc(1);
Yo=Yo + delc(2);
Zo=Zo + delc(3);
rdt=rdt + delc(4);
disp_del=[iter; delc(1); delc(2); delc(3); delc(4)]
del=delc(1:4);

```

```

v=f-B*del;
phi=v'*W*v;
if(abs(phi-old_phi)/phi < threshold)
    disp('we have converged');
    converged=1;
    break;
end

old_phi=phi;
end

if(converged == 0)
    disp('we did not converge');
else
    % we converged
    % show results
    disp('receiver location (km)');
    [Xo Yo Zo]
    disp('receiver clock bias (usec)');
    [rdt]
end
disp('residuals (km)');
v
rms=sqrt(v'*v/nobs)
disp('constraint misclosure (meters)');
constr_misc=0.100 - sqrt((Xo-Xf)^2 + (Yo-Yf)^2 + (Zo-Zf)^2);
constr_misc_m=constr_misc*1000

% make global test and post adjustment statistics
% careful - you must modify redundancy if you reject satellites
% low on the horizon

r
test_stat=v'*W*v/(sigma0^2)
alpha = 0.05
cv1=icdf('chi2',alpha/2,r)
cv2=icdf('chi2',1-alpha/2,r)
if((test_stat > cv1) && (test_stat < cv2))
    %Sdd=sigma0^2 * Qdd;
    pass=1;
    disp('pass global test at alpha = 0.05');
else
    sigma0_hat_sqr=v'*W*v/r;
    %Sdd=sigma0_hat_sqr * Qdd;
    pass=0;
    disp('fail global test at alpha = 0.05');
end

```